

AXIAL VIBRATION OF ELASTICALLY SUPPORTED BEAMS

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ABSTRACT

In this study free axial vibration of an elastically supported beam is analyzed. The supports are modeled by elastic translational springs. The frequency values for the first three vibration modes of the beam are obtained for various values of spring constants and presented in the tables. The frequency values for the spring constants of zero and infinity are also compared, respectively, with the ones of free and fixed beams and nearly the exact values are obtained with negligible error percentages.

Keywords: Axial vibration, Beam, Elastic Support, Frequency

1. Introduction

In practice, the representation of a beam by a discrete model is an idealized model; however, in fact, beams have continuously distributed mass and elasticity. Mostly, especially for the axially vibration, beams are modeled as continuous systems having infinite number of degreed of freedom [1-6].

In this study, the free vibration analysis of an elastically supported axially vibrating beam is made. The elastic springs against translation are used to model the supports. The differential equation of motion of the axially vibrating beam is solved by separation of variables method [7] and the displacement function is obtained. The boundary conditions are written for the elastic supports. The natural frequencies for the first three modes are obtained for the various values of the spring constants. The results obtained for the spring constant value of zero are compared with the frequency values of free beam whereas the ones for the spring constant value of infinity are compared with the frequency values of fixed beam. In addition, the frequencies obtained for the left end spring constant value of infinity and the

right end spring constant value of zero are compared with the frequency values of cantilever beam. The axially vibrating beam considered in the study is assumed to be homogeneous and isotropic.

2. Solution of Equation of Motion for an Axially Vibrating Beam

An axially vibrating beam, given in Figure.1, with the distributed mass m, the length L, the modulus of elasticity E, the cross-section area A and the axial rigidity AE has a dimensional differential equation of motion for free vibration as [8]

$\xrightarrow{u(x,t)}$	N 11
m, L, AE	─ ····>>x

Figure 1: An Axially Vibrating Beam

$$\frac{\partial^2 u(x,t)}{\partial x^2} - \frac{m}{AE} \frac{\partial^2 u(x,t)}{\partial t^2} = 0$$
(1)

where u(x,t) is the displacement function of the beam in terms of both displacement x and time t. Application of the separation of variables method to Eq. (1) as in the form of Eq. (2) is commonly used in vibration analysis of beams.

$$u(x,t) = X(x) \cdot T(t) = X(x) \cdot [A \cdot \sin(\omega t) + B \cdot \cos(\omega t)]$$
(2)

In Eq. (2), X(x) is the eigenfunction named as shape function, T(t) is time function, ω is the eigenvalue of the solution named as natural frequency and A, B are constants.

The derivatives used in Eq. (1) can, therefore, be written as

$$\frac{\partial^2 u(x,t)}{\partial x^2} = u^{''}(x,t) = X^{''}(x) \cdot [A \cdot \sin(\omega t) + B \cdot \cos(\omega t)] = X^{''}(x) \cdot T(t)$$

$$\frac{\partial^2 u(x,t)}{\partial t^2} = \ddot{u}(x,t) = X(x). (-\omega^2)[A.\sin(\omega t) + B.\cos(\omega t)] = -\omega^2.X(x).T(t)$$
(4)

where $\binom{n}{2}$ and $(\ddot{})$ denote the second order derivative due to x and t, respectively. Substitution of Eq. (3) and Eq. (4) in Eq. (1) gives the governing equation of motion in the form as

$$X''(x).T(t) + \frac{m\omega^{2}}{AE}X(x).T(t) = 0 \quad X''(x) + \frac{m\omega^{2}}{AE}X(x) = 0 \quad 0 \le x \le L$$

(5)

for
$$\alpha^2 = \frac{m\omega^2}{AE}$$
 $X''(x) + \alpha^2 X(x) = 0$

(6)

The characteristic equation and the solution of Eq. (6) is given as follows as D being d/dz:

$$D^2 + \alpha^2 = 0 \quad \rightarrow \quad D_{1,2} = \pm i\alpha \tag{7}$$

$$X(x) = C_1 \sin(\alpha x) + C_2 \cos(\alpha x)$$
(8)

Eq. (8) gives the shape function of the axially vibrating beam due to the displacement variable, x. Therefore, from Eq. (2), the displacement function of the axially vibrating beam has the form of Eq. (9).

$$u(x,t) = X(x).T(t) = [C_1.\sin(\alpha x) + C_2.\cos(\alpha x)].T(t)$$
(9)

3. Boundary Conditions



Figure 2: Elastically Supported Axially Vibrating Beam

Two boundary conditions have to be written for the elastically supported beam in Figure.2 since two integration constants (C_1 , C_2) are obtained in the solution of second order differential equation of motion. The boundary conditions written for the left and the right ends of axially vibrating beam are given, respectively, as [9]

for x=0
$$N(x = 0, t) = AE \frac{\partial u(x=0,t)}{\partial x} = AEu'(x = 0, t) = k_0 \cdot u(x = 0, t)$$

for x=L
$$N(x = L, t) = AE \frac{\partial u(x=L,t)}{\partial x} = AEu'(x = L, t) = -k_L \cdot u(x = L, t)$$
(11)

where k_o and k_L are the spring constant values of, respectively, the left end and the right end supports, N(x,t) is the axial force. If Eq. (9) and its derivative are substituted into Eq. (10) and Eq. (11) one gets the following relation between the coefficient matrix and the integration constants.

$$\begin{bmatrix} k_{11} & k_{12} \\ k_{21} & k_{22} \end{bmatrix} \cdot \begin{pmatrix} C_1 \\ C_2 \end{pmatrix} = \begin{pmatrix} 0 \\ 0 \end{pmatrix} \rightarrow [k] \cdot \begin{pmatrix} C_1 \\ C_2 \end{pmatrix} = \{0\}$$
(12)
$$|k| = \begin{vmatrix} k_{11} & k_{12} \\ k_{21} & k_{22} \end{vmatrix} = 0$$
(13)

where [k] is the coefficient matrix and $k_{11} = \propto$, $k_{12} = \frac{-k_0}{AE}$, $k_{21} = \propto .\cos(\propto L) + \frac{k_L}{AE} .\sin(\propto L)$, $k_{22} = -\propto .\sin(\propto L) + \frac{k_L}{AE} .\cos(\propto L)$. For non-trivial solution equating the determinant of the coefficient matrix of Eq. (12) to zero, as in Eq. (13), will give the eigenfrequencies of the axially vibrating beam with elastic supports. These frequencies are computed by a program written by the author considering the secant method [10].

4. Numerical Analysis

The first three natural frequencies of the axially vibrating beam with elastic supports are calculated for the k_0 and k_L values of 0, 10^3 , 10^6 , 10^9 , 10^{12} and 10^{18} , the beam length of L=1 m. and the modulus of elasticity of E=2100000 kg/cm². IPB-100, IPB-300 and IPB-600 profiles are used for numerical analysis with the mechanical properties given in Table 1 where h is height, G is weight per length, A is cross-section area and AE is axial rigidity of the corresponding profile. The distributed mass of the beam m is calculated from G/g as g being the acceleration of gravity with the value of 981 cm/sn².

Profile	h	G	Α	AE
Tionic	(cm)	(kg/cm)	(cm ²)	(kg)
IPB100	10	0.081	10.3	21630000
IPB300	30	0.422	53.8	112980000
IPB600	60	1.22	156	327600000

Table 1: The Mechanical Properties of the Profiles Used in This Study

The frequency values computed due to different values of the spring constants for the both ends are presented in Table 2, Table 3 and Table 4 for, respectively, IPB-100, IPB-300 and IPB-600.

Table 2: Frequencies Computed due to Different Values of the Spring Constants for IPB-100

k ₀ =k _L	10³	10 ⁶	10 ⁹	10 ¹²	10 ¹⁵	10 ¹⁸
ω ₁	492	11462	16073	16079.403691	16079.410640	16079.410647
W ₂	16095	24039	32145	32158.807382	32158.821280	32158.821294
ω ₃	32167	37876	48218	48238.211074	48238.231921	48238.232041

Table 3: Frequencies Computed due to Different Values of the Spring Constants for IPB-300

k ₀ =k _L	10³	10 ⁶	10 ⁹	10 ¹²	10 ¹⁵	10 ¹⁸
ω ₁	216	6355	16064	16100.077062	16100.113406	16100.113442
ω_2	16104	18558	32128	32200.154125	32200.226812	32200.226884
ω3	32202	33577	48192	48300.231187	48300.340217	48300.340327

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$k_0 = k_L$	10 ³	10 ⁶	10 ⁹	10 ¹²	10 ¹⁵	10 ¹⁸
ω ₁	127	3912	16020	16124.028685	16124.134224	16124.134330
ω ₂	16126	17064	32039	32248.057370	32248.268448	32248.268659
ω3	32249	32740	48058	48372.086055	48372.402672	48372.402989

Table 4: Frequencies Computed due to Different Values of the Spring Constants for IPB-600

In Table 5, the frequency values obtained for the spring constant values of zero and infinity ($k_0=k_L=10^{18}$ in this study) are compared with the frequency values obtained for either free or fixed beams that have the same frequency equation.

Table 5: Comparison of Frequencies Computed for $k_0=k_L=0$ and $k_0=k_L=\infty$ (10¹⁸ in this study) with the Frequencies of Free or Fixed Beam

		$k_0 = k_L = 0$	$\omega = \pi \frac{\pi}{AE}$
		$k_0 = k_L = \infty$	$\omega_i = n_i \frac{1}{L} \sqrt{m}$
	ω1	16079.410647	16079.410680
IPB-100	ω ₂	32158.821294	32158821360
	ω3	48238.231941	48238.232041
	ω1	16100.113442	16100.113434
IPB-300	ω2	32200.226884	32200.226868
	ω3	48300.340327	48300.340302
	ω1	16124.134330	16124.134330
IPB-600	ω ₂	32248.268659	32248.268660
	ω3	48372.402989	48372.402990

The frequency values computed due to different values of the spring constant k_0 and $k_L=0$ are presented in Table 6, Table 7 and Table 8 for, respectively, IPB-100, IPB-300 and IPB-600.

k ₀ =k _L	10³	10 ⁶	10 ⁹	10 ¹²	10 ¹⁵	10 ¹⁸
ω ₁	348	6640	8038	8039.703585	8039.705322	8039.705324
ω ₂	16087	20470	24114	24119.110754	24119.115966	24119.115971
ω3	32163	35189	40190	40198.517923	40198.526609	40198.226618

Table 6: Frequencies Computed due to Different Values of k_0 and $k_L=0$ for IPB-100

Table 7: Frequencies Computed due to Different Values of k_0 and $k_L=0$ for IPB-300

$\mathbf{k_0} = \mathbf{k_L}$	10³	10⁶	10 ⁹	10 ¹²	10 ¹⁵	10¹⁸
	153	4214	8041	8050.047626	8050.056712	8050.056721
	16102	17407	24123	24150.142878	24150.170136	24150.170163
ω_2	10102		-	24130.142070	24130.170130	24130.170105
ω3	32201	32903	40205	40250.238131	40250.283560	40250.283605

Table 8: Frequencies Computed due to Different Values of k₀ and k_L=0 for IPB-600

$k_0 = k_L$	10 ³	10 ⁶	10 ⁹	10 ¹²	10 ¹⁵	10¹⁸
ω ₁	90	2700	8036	8062.040754	8062.067138	8062.067165
ω ₂	16125	16607	24108	24186.122261	24186.201415	24186.201495
ω3	32249	32496	40179	40310.203768	40310.335692	40310.335824

In Table 9, the frequency values obtained for the spring constant values of $k_0=\infty$ and $k_L=0$ are compared with the frequency values obtained from the frequency equation of cantilever beam.

Cantilever Beam							
		$k_0 = \infty$	$(2n_i-1)\pi AE$				
		k _L =0	$\omega_i = \frac{(2n_i - 1)}{2} \frac{\pi}{L} \sqrt{\frac{AE}{m}}$				
	ω ₁	8039.705324	8039.705340				
IPB-100	ω ₂	24119.115971	24119.116020				
	ω3	40198.226618	40198.226700				
	ω1	8050.056721	8050.056717				
IPB-300	ω2	24150.170163	24150.170151				
	ω3	40250.283605	40250.283585				
	ω1	8062.067165	8062.067165				
IPB-600	ω2	24186.201495	24186.201495				
	ω3	40310.335824	40310.335825				

Table 9: Comparison of Frequencies Computed for $k_0\!\!=\!\!\infty$ and $k_L\!\!=\!\!0$ with the Frequencies of

5. Conclusions

In this study free longitudinal vibration of an elastically supported beam is made. The natural frequency values are obtained for different values of spring constants at both ends and presented in tables. It can be seen from Tables 2, 3, 4 and Tables 6, 7 and 8 that as the spring constant values increase through a value of 10^9 the frequency values rapidly increase, however, at the value of 10^9 the frequency values are so close to its ideal limit obtained from the frequency equation of ideal support condition, being free or fixed. As the spring constant

value increases from 10^9 to theoretically infinity (practically 10^{18} in this study) the frequency values show gentle increase and at the spring constant value that represents infinity the frequency value reaches its limit value for the considering support. Increasing the height of the beam section causes a decrease in frequency values for the spring constant values of less than 10^9 and an increase for the spring constant values from 10^9 to infinity at which the frequency values show an increase no longer.

References

- 1. A.J. Wilson and P.J. Myers, The Longitudinal Vibrations of an Elastic Cylinder Under Large Axial and Normal Tractions, *International Journal of Solids and Structures*, 29(24), 1992, 3153-3168.
- Q.S. Li, G.Q. Li and D.K. Liu, Exact Solutions for Longitudinal Vibrations of Rods Coupled by Translational Springs, *International Journal of Mechanical Scienses*, 42(6), 2000, 1135-1152.
- 3. B.M. Kumar and R.I. Sujith, Exact Solutions for the Longitudinal Vibration of Non-Uniform Rods, *Journal of Sound and Vibration*, 207(5), 1997, 721-729.
- M. Rumerman and S. Raynor, Natural Frequencies of Finite Circular Cylinders in Axially Symmetric Longitudinal Vibration, *Journal of Sound and Vibration*, 15(4), 1971, 529-543.
- W. Surnelka, R. Zaera and J. Fernandez-Saez, A Theoretical Analysis of the Free Axial Vibrations of Non-Local Rods with Fractional Continuum Mechanics, *Meccanica*, 2015, in press.
- O. Demirdag and B. Yildirim, Dimensionless Analysis of Axially Vibrating Beams, Journal of Multidisciplinary Engineering Science and Technology, 2(8), 2015, 2207-2209.
- O. Demirdag, Free Vibration of Elastically Supported Timoshenko Columns with Attached Masses by Transfer Matrix and Finite Element Methods, *Sadhana*, 31(1), 2008, 57-68.
- 8. J.J. Tuma and F.Y. Cheng, *Theory and problems of dynamic structural analysis* (Schaum's Outline Series, USA: McGraw-Hill, Inc., 1983).

- 9. V. Yerlici and H. Lus, *Introduction to structural dynamics* (Istanbul, Bogazici University Publishing, Turkey, 2007 (in Turkish)).
- 10. K.H. Low, A Comprehensive Approach for the Eigenproblem of Beams With Arbitrary Boundary Conditions, *Computers & Structures*, *39*, 1991, 671-678.